

# **Numerical and experimental analysis of the structural performance of AM components built by Fused Filament Fabrication**

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## **ABSTRACT**

This work analyses the performance of parts built by Additive Manufacturing (AM) using Fused Filament Fabrication (FFF) demonstrating the correlation between the printing orientation and structural performance. FFF components present two regions showing different mechanical behaviour: the external contour and the inner structure (in-fills or lattice). The respective mechanical properties of the contour and the inner structure are obtained. In this work the inner structure is replaced by an anisotropic homogenized material. A Representative Volume Element (RVE) with Periodic Boundary conditions (PBC) is adopted to obtain the corresponding equivalent constitutive tensor. On the other hand, the contour is considered isotropic. The material characterization of both the in-fill and the contour is done following two complementary strategies: (i) an experimental campaign involving several tensile tests on FFF specimens; (ii) a sensitivity analysis through numerical modelling. Performing experiments to obtain the material properties for contour and in-fill may be a challenging task. Thus, the numerical modelling and the optimization technique are used to obtain the material properties as a function of the filament properties. Calibration of the structural response of 3D-printed demonstrators under bending and torsion is done in order to optimize the material parameters of the numerical model by minimising the difference between the experimental and numerically computed structural stiffness. It is shown that assuming isotropic behaviour for the contour results in a negligible error. The AM software so calibrated can be used for analysing the mechanical performance of FFF components and selecting the optimal printing orientation as well as the contour thickness and in-fill density to satisfy the structural performance required.

**Keywords:** Additive Manufacturing; Fused Filament Fabrication; Representative Volume Element; Homogenization; Anisotropy.

## 1 Introduction

Additive Manufacturing (AM), also known as 3D printing is a technique to fabricate parts and components from a Computer-Aided Design (CAD) model. This is done through layer-by-layer deposition of different materials ranging from polymers to metals, as well as cement-based compounds. The interest in AM technology grows in both the industrial and the scientific communities due to the fabrication versatility it provides, compared to traditional manufacturing technologies. AM has a particularly strong advantage in terms of its capacity of creating complex geometries with cavities or lattice structures. Moreover, AM reduces the assembly time, as it allows part consolidation. These characteristics make AM a powerful cost-effective technology allowing for higher design freedom and customization compared to manufacturing methods such as forging or casting [1, 2].

Polymer-based 3D-printing is evolving fast, moving the focus from rapid-prototyping to the fabrication of structural components with enhanced thermal resistance as well as mechanical strength while reducing weight and material use [3]. Acrylonitrile butadiene styrene (ABS), polycarbonate (PC) and polylactic acid (PLA) are some examples of polymer-based materials commonly used in AM processes [4].

There are different types of AM techniques where polymer-based materials are commonly used: *Selective Laser Sintering* (SLS) [5, 6]; *Multi Jet Fusion* (MJF) [7, 8] and *Fused Filament Fabrication* (FFF), also known as *Fused Deposition Modelling* (FDM) [9, 10, 11].

Generally in AM, the material properties of the final printed component do not coincide with the ones of the raw material [12]. This occurs due to the layer-by-layer fabrication, characteristic of the AM processes. Moreover, the mechanical performance depends on the resulting microstructure, which is influenced by the local temperature gradients aligned with the building direction.

While some AM types such as SLS or MJF maintain the isotropy of the raw material, in components printed via FFF anisotropic behaviour is observed. Despite the fact that the underlying filament material is isotropic, the anisotropy of a FFF-printed part manifests due to: the layer-to-layer adhesion, the filament-to-filament adhesion and the printing patterns.

The significant material anisotropy and heterogeneity encountered in FFF parts cannot be ignored as it has a significant impact on the mechanical performance of the component [13]. This is the reason why establishing a methodology for the material characterization of components fabricated by FFF technology is one of the main focus of the present work. In FFF, the objects are printed by melting a thermoplastic filament to be extruded through a nozzle. The printing process and, consequently, the final structural behaviour are characterized by different process parameters, the most important ones being the object orientation, the layer thickness and the scanning patterns.

These parameters are defined by the user prior to the slicing of the CAD model. Selecting the optimal process parameters in FFF is critical as this directly affects: (i) the mechanical behaviour of the component [14], (ii) its surface texture [15] and, (iii) the fabrication cost [16].

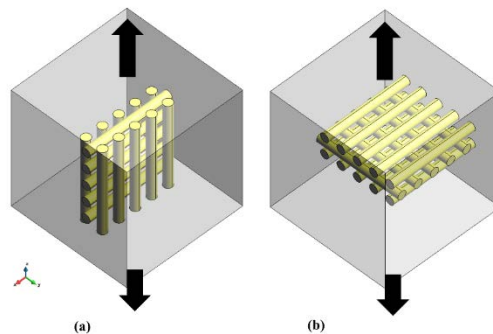
The printing patterns can be classified into contour and raster trajectories. The contours define the exterior surface of the 3D object, while both the raster and the in-fill density characterize the inner section bounded by the contour. There are two types of inner structures: the *in-fill* structures and the *lattice* structures (Figure 1). The in-fill is the standard structure adopted in FFF technology and its density is defined by the raster to raster air gap. The in-fill pattern is repeated for each layer following the printing (vertical)

direction. The lattice structures are characterized by a *unit cell* periodically repeated in space. Both structures are used to save material, printing time and production costs.

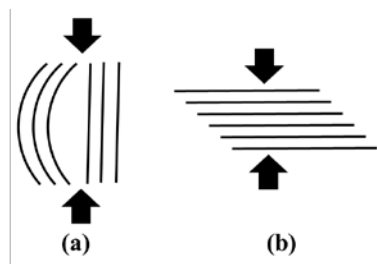


**Figure 1. Types of inner structure: in-fill (left), lattice (right)**

Figure 2 shows a geometry fabricated with the same printing parameters. Nevertheless, when the loading is aligned with the filament direction (Figure 2(a)) the part shows higher stiffness and strength than when loading is applied orthogonal to the printing plane (Figure 2(b)). Therefore, in case of tensile stresses, the mechanical performance of FFF components improves when the filaments are aligned with the loading [17]. Contrariwise under compression it is preferable to align the loading with the stacking direction to avoid buckling (Figure 3(a)). However, increasing the compression, sliding between layers due to shear forces can affect the structural integrity (Figure 3(b)).



**Figure 2. Filaments pattern depending on the printing orientation: (a) Printing orientation parallel to the loading force; (b) Printing orientation perpendicular to the loading force.**



**Figure 3. Scheme of the compressive loading application (a) Buckling (b) Shearing**

Numerous experiments have reported the strong impact of the component orientation and the raster angle on the properties of the printed parts [14, 18-25]. However, only a few studies are devoted to the numerical simulation of the mechanical performance of FFF builds. Casavola et al [26] characterized the orthotropic properties of the FFF parts printed by ABS and PLA utilizing Classical Lamination Theory (CLT). Garg and Bhattacharya [27] simulated the deformation of the FFF samples under uniaxial loading adopting different raster thicknesses and printing directions. In the majority of the numerical simulation studies, including the above-mentioned ones, FFF components are

modelled as a single solid material. However, structural response of contours and inner structures is different, depending on the corresponding building pattern and in-fill density. In this work, we propose a methodology for characterizing the mechanical performance of components manufactured by FFF technology. This methodology relies on accounting for the mechanical properties of the contours and the inner structures, separately. Both an experimental campaign and a material properties calibration through numerical modelling are carried out. The corresponding mechanical properties of the contour and the in-fill are acquired.

On the experimental side, several dog-bone specimens are manufactured to perform the uniaxial tensile tests in order to characterize the anisotropic behaviour of the printed material. Next, a door-handle component is fabricated and tested under bending and torsion loading to investigate the influence of the printing orientation, the layer thickness and the in-fill parameters on the actual structural performance. The measured values of both the stiffness and strength shown by these demonstrators are used for the calibration of the numerical model.

On the numerical side, a homogenization technique using the Representative Volume Element (RVE) is used to characterize the inner (lattice and in-fill) structures. The equivalent anisotropic material behaviour is considered for the inner structures while the contour is assumed to be isotropic.

Carrying out experimental tests to determine the material properties for contour and in-fill may be a challenging task. However, the numerical modelling and optimization technique provide an alternative mean for the material characterization. Therefore, an in-house FE software platform is customized to enhance the design of structural components built by FFF.

The outline of the paper is the following: first, the experimental set-up used in this work is presented in Section 2. This includes the experimental campaign on different specimens. In Section 3, the numerical strategy for material characterization is described. This includes: the constitutive model (Section 3.2), the homogenization technique to characterize the in-fill structures (Section 3.3 and 3.4) and the sensitivity analysis performed to calibrate the AM model (Section 3.5). Finally, in Section 4, the results are discussed and future research lines stemming out of the present work are presented.

## 2 EXPERIMENTAL PROCEDURE

This Section describes the experimental procedure carried out for the characterization of the printed material, as well as the tests performed on the demonstrators. The former ones are to analyse the mechanical behaviour of the samples with printing patterns of the external contour. The latter ones are used for the calibration of the numerical model developed and to analyse the mechanical behaviour of a complex structure under different loading conditions.

All the specimens and the demonstrators analysed in the present work were printed and tested by EURECAT *Centre Tecnològic de Catalunya*. The Original Prusa i3 MK2S is the FFF printer used for the manufacturing. *Ultimaker Cura*® is the slicing software used to define the printing parameters of the parts [28]. The printing material is PLA extruded at 215°C. Linear in-fill (at 45°) is chosen for all the specimens. Table 1 summarizes the printer information.

**Table 1. Printer information**

Tester	EURECAT
3D printer	Original Prusa i3 MK2S
Printer technology	FFF
Software	Ultimaker Cura
Material	PLA
Extrusion Temp.	215°C
Type of in-fill	Rectilinear

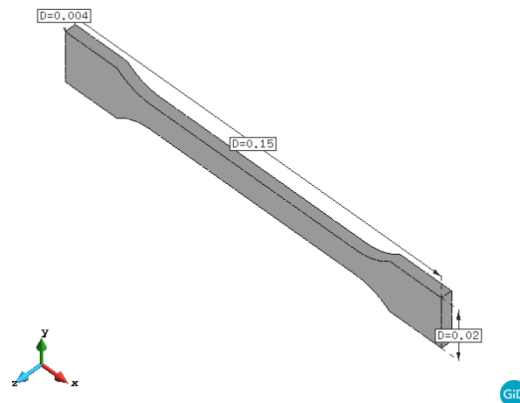
## 2.1 MATERIAL CHARACTERIZATION BY EXPERIMENTS

As motivated in the introduction, in this work, the structural behaviour of the external contours and inner structures (in-fill and lattice) are distinguished. For the contour, there exist two main printing patterns used in FFF: (1) aligned filaments and, (2)  $\pm 45^\circ$  crossed patterns. The former is used for most of the surfaces when their slope with respect to the printing plane (XY) is greater than  $30^\circ$ . The latter is typically used for the top and bottom covers where the surfaces are almost horizontal (slope less than  $30^\circ$ ).

The objective of the experimental work is to compare the material properties of the raw material (filament) with the behaviour of FFF-printed specimens using the above mentioned printing patterns.

On the one hand, the single filament (PLA) is homogeneous and isotropic, and the corresponding elastic parameters are: Young's modulus = 3368 MPa and Poisson's ratio=0.35 [29].

On the other hand, dog-bone specimens are printed directly from the CAD model (see Figure 4) and the corresponding material behaviour is assumed to be orthotropic and aligned with the material directions of the selected printing pattern. Therefore, to fully determine the orthotropic stiffness matrix, nine constants must be calculated: three Young's moduli, three Poisson's coefficients and three shear moduli according to the three directions X, Y and Z. In order to obtain these 9 independent parameters, several specimens are printed in 6 different orientations (see Figures 5, 7 and 9) and tested (uniaxial tensile tests) according to the Standard Test Method for Tensile Properties of Plastics (ASTMD638). The cube in figures 5, 7 and 9 is shown only for the sake of visualizing the 6 printing orientations.



**Figure 4. Geometry of the dog-bone specimens**

According to Hooke's law of elasticity, the uniaxial tensile test of a specimen printed according to the orientation 1, defines the Young's modulus in direction X and the Poisson's ratio in XY plane as:

$$E_x = \frac{\sigma_{xx}}{\varepsilon_{xx}}, \nu_{xy} = -\frac{\varepsilon_{yy}}{\varepsilon_{xx}} \quad (1)$$

Similarly, from orientation 2, one obtains:

$$E_y = \frac{\sigma_{yy}}{\varepsilon_{yy}}, \nu_{yz} = -\frac{\varepsilon_{zz}}{\varepsilon_{yy}} \quad (2)$$

and from orientation 3:

$$E_z = \frac{\sigma_{zz}}{\varepsilon_{zz}}, \nu_{xz} = -\frac{\varepsilon_{xx}}{\varepsilon_{zz}} \quad (3)$$

Young's modulus and Poisson's ratio can be obtained by a stress / elongation test with equation 1-3 relating the direction of application of the force or displacement to the one perpendicular to it.

From orientations 4, 5 and 6, the shear moduli are obtained by uniaxial tensile tests of specimens printed at 45° and belonging to the corresponding XY, XZ and YZ printing planes. Thus, the following relationships are used to obtain the shear modulus within each plane ([14], [18]):

$$G_{xy} = \frac{E_1}{2(1+\nu_{12})} \Big|_{xy}, G_{xz} = \frac{E_1}{2(1+\nu_{12})} \Big|_{xz}, G_{yz} = \frac{E_1}{2(1+\nu_{12})} \Big|_{yz} \quad (4)$$

where, in this case, 1 stands for the direction of the applied load and 2 is the direction perpendicular to 1.

Considering the above, 3 different families of FFF-printed specimens are printed and tested:

1. Crossed-type: 30 specimens (5 for each orientation) characterized by ±45° crossed printing pattern (see Figure 5).
2. Mixed-type: 30 specimens (5 for each orientation) characterized by 5 aligned filaments following the external contour of the dog-bone shape. The rest of the volume is filled by by ±45° crossed printing pattern (see Figure 7).
3. Aligned-type: 30 specimens (5 for each orientation). In this case, 5 aligned filaments following the external contour are printed, only. Each filament is 0.4 mm tick (see Figure 9).

Figure 11 shows the contour and in-fill in all the families printed in orientation 1. Table 2 summarises the printing patterns of each family. The samples are printed and tested according to the ISO 527-2 1A.

**Table 2. Printing patterns of each family**

Family	pattern
1	Crossed
2	Mixed
3	Aligned

The stress/strain curves obtained for the 3 families and the 6 printing orientations are shown in Figure 6, Figure 8 and Figure 10, respectively.

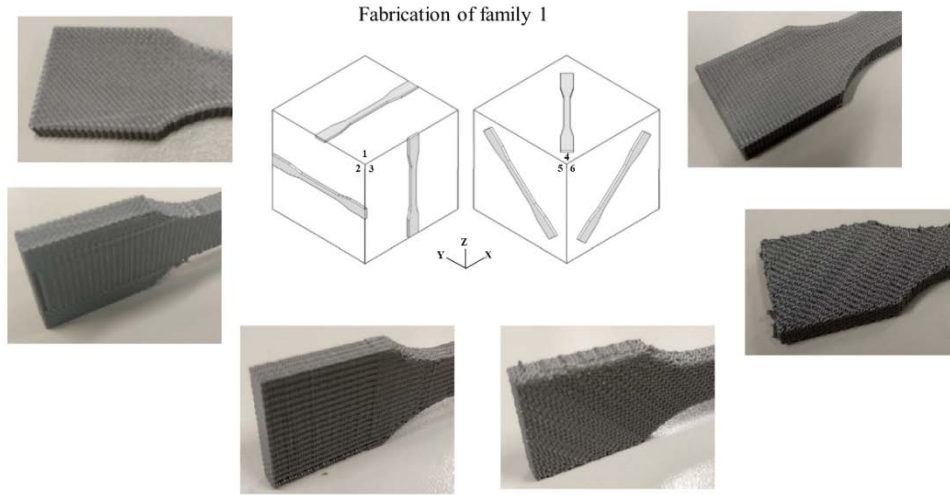
The corresponding mechanical properties are presented in Table 3.  $E_x$  and  $\nu_{xy}$  are obtained from specimens printed and tested in plane 1 (Plane XY). The aligned-type specimens of orientations 1 and 4 could not be experimentally tested because they have no interior and the tests fail.

Note that the ratio between aligned and crossed pattern of the mixed-type specimens depends on their actual orientation. In particular, only orientations 1 and 4 present mixed

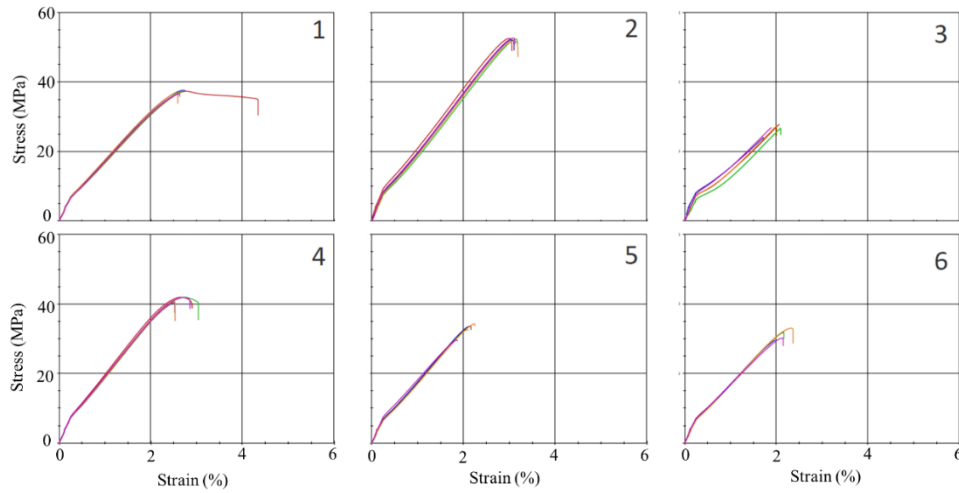
printing patterns while for orientations 2, 3, 5 and 6 the aligned filaments occupy the whole thickness of the dog-bone specimens. Therefore, the elastic constants obtained using these orientations are all equal for families 2 and 3.

**Table 3. Mechanical properties obtained from the tensile test, values are in GPa.**

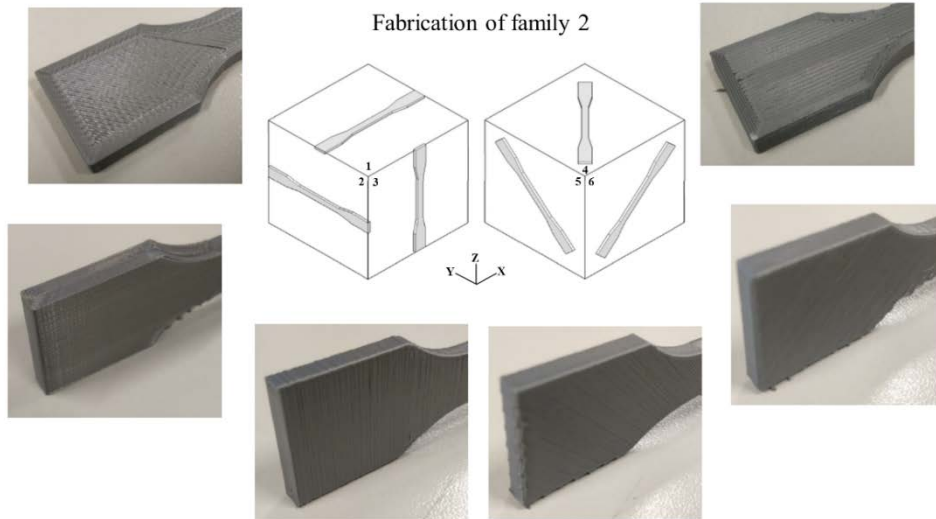
Pattern	$E_x$	$E_y$	$E_z$	$\nu_{xy}$	$\nu_{yz}$	$\nu_{xz}$	$G_{xy}$	$G_{yz}$	$G_{xz}$
Crossed	2.464	1.984	2.324	0.259	0.188	0.236	0.332	0.362	0.363
Mixed	2.772	3.171	3.058	0.227	0.221	0.181	0.363	0.340	0.341
Aligned	-	3.171	3.058	-	0.221	0.181	-	0.340	0.341



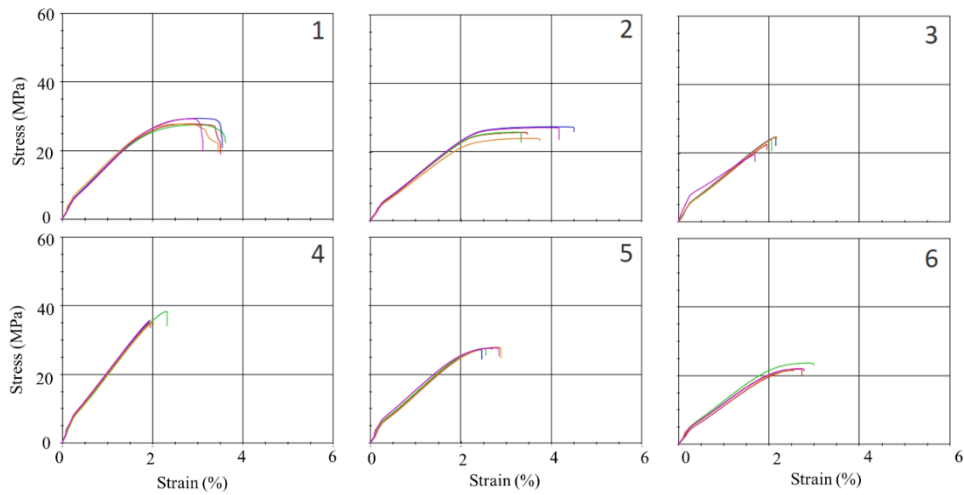
**Figure 5. Crossed-type: 6 printing orientations**



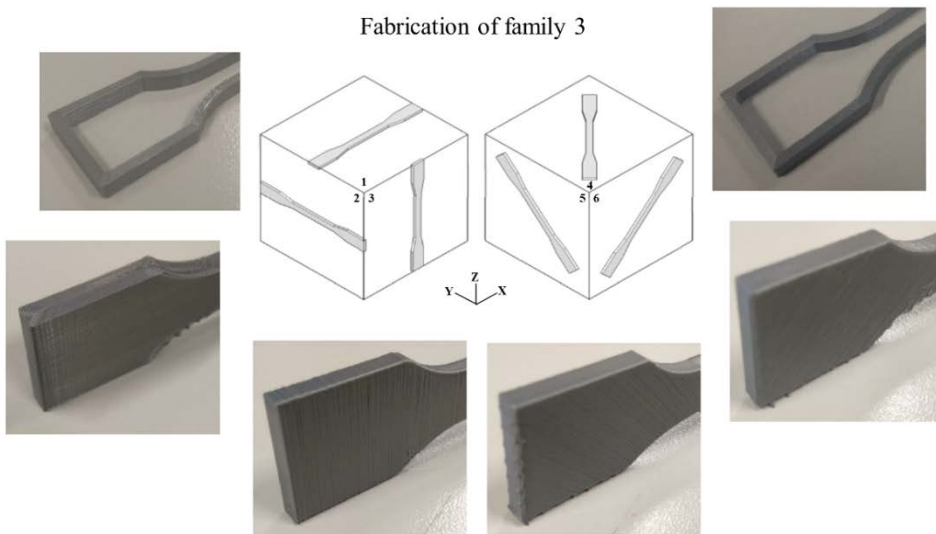
**Figure 6. Stress-strain curves of crossed printing pattern**



**Figure 7. Mixed-type: 6 printing orientations**

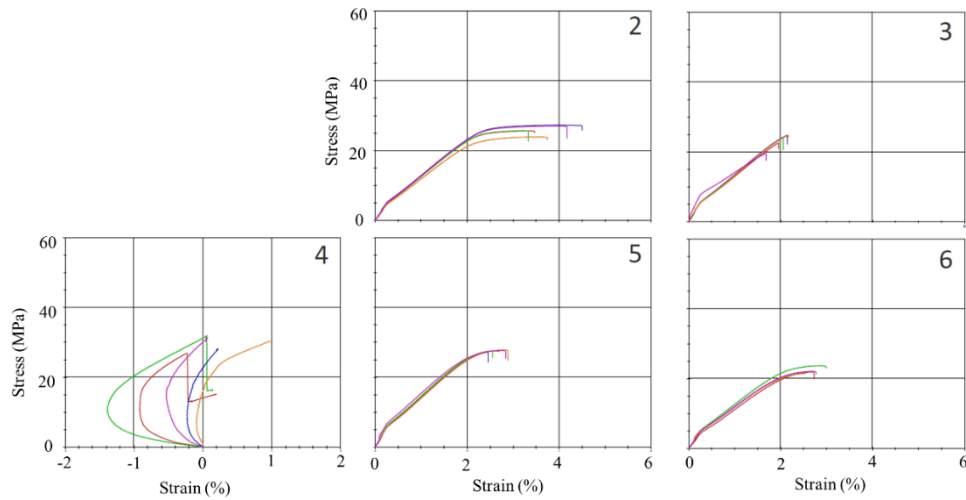


**Figure 8. Stress-strain curves of mixed printing pattern**

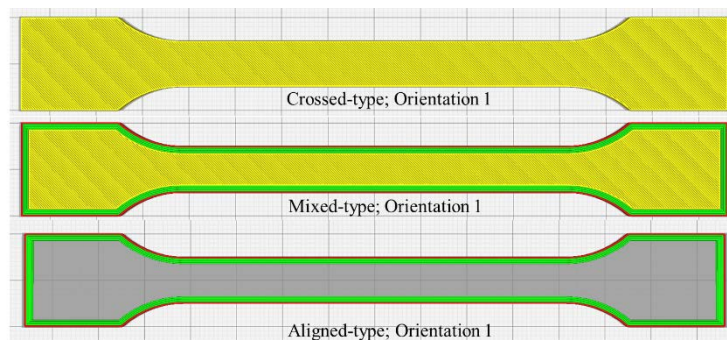


**Figure 9. Aligned-type: 6 printing orientations**





**Figure 10. Stress-strain curves of aligned printing pattern**



**Figure 11. Crossed, mixed and aligned printing patterns in orientation 1: contour (green) and in-fill (yellow)**

This experimental work shows that the mechanical properties of FFF-printed specimens depend on the actual printing pattern. Moreover, it can be concluded that:

- (i) the highest stiffness is achieved when the specimens are fabricated with aligned filaments. Nevertheless, the Young's modulus is around 10-15% lower than for the raw material (single filament).
- (ii) In the case of crossed-type specimens, the material stiffness shows values 30-35% lower than that of the raw material. It must be also noted that the lack of a smooth external contour covering the external skin of the specimens generates stress concentration points where cracking can occur.
- (iii) Some of the orientations are extremely difficult to be tested (e.g. orientations 1 and 4 in Figure 10).

### 3 NUMERICAL STRATEGY

In this Section a numerical strategy for predicting the structural performance of components fabricated by FFF is presented. The model is initially fed with the material parameters obtained from the experimental work realized on the dog-bone specimens. Next, these parameters are calibrated through a sensitivity analysis to accurately capture the mechanical response of a demonstrator fabricated in 3 different orientations.

### 3.1 MECHANICAL ANALYSIS OF FFF PIECES

In the structural components fabricated by FFF shown in Figure 1, two regions are distinguished: the external contours and the inner part; either lattice or in-fill structures. Focussing on the external contours, the experimental work carried out demonstrates that: (i) the as-printed material is characterized by a lower material stiffness and strength if compared to the raw material; (ii) the printing pattern plays an important role on the material characterization. The mechanical behaviour of the aligned-type and crossed-type samples are transversely isotropic. Depending on the printing pattern, the isotropic plane varies. The crossed-type samples are transversely isotropic where XZ is the isotropic plane with the stiffness lower in Y direction. The mixed-type and aligned-type samples are also transversely isotropic. However, YZ is the isotropic plane and the stiffness is lower in X direction. When modelling industrial components defined by complex geometries, the actual material anisotropy of the printed material becomes a challenging task due to the alignment of the filaments with the external surface contour. Moreover, it is not always easy to split this contour by different printing patterns. Thus, in the numerical model, the external contour is assumed as an *isotropic* bulk material, and the *mixed-type* material parameters obtained from Family-2 are used as the starting values for the numerical modelling.

The most standard inner structure in FFF is characterized by an alternating sequence of +45° and -45° rasters piled up in the building (vertical) direction, being the in-fill density, a printing parameter defining the raster-to-raster spacing.

The in-fill structure is heterogeneous and its discretization by the standard FE mesh would result in an excessive number of elements, precluding efficient numerical computations. Alternatively, it can be modelled as an *equivalent* homogeneous continuum taking advantage of its repetitive cell structure. Hence, in the present work, a homogenization method is adopted to characterise the corresponding *anisotropic* elastic behaviour.

### 3.2 CONSTITUTIVE MODELLING OF THE 3D-PRINTED COMPONENTS

The aim of this work is to perform characterization of the structural *stiffness* of AM components built by FFF. The constitutive behaviour will be characterized by a linear elastic model:

$$\boldsymbol{\sigma}_m = \mathbf{C}_m : \boldsymbol{\varepsilon}_m \quad (5)$$

where  $\boldsymbol{\sigma}_m$  and  $\boldsymbol{\varepsilon}_m$  are the Cauchy stress and strain tensors, respectively.  $\mathbf{C}_m$  is the constitutive (stiffness) tensor of the material. Alternatively, the constitutive relationship can be written as:

$$\boldsymbol{\varepsilon}_m = \mathbf{D}_m : \boldsymbol{\sigma}_m \quad (6)$$

where  $\mathbf{D}_m = \mathbf{C}_m^{-1}$  is the compliance (flexibility) tensor, defined as the inverse of the stiffness tensor.

The *transversely isotropic* constitutive model is suitable for the *inner structures* and, particularly, for the *in-fill structures* fabricated by FFF technology. Its use is justified thanks to the periodic arrangement of the unit cells and the printing evolution in the building direction. Being XY the symmetry plane (built plate) and Z, the building direction, the flexibility tensor of transversely isotropic materials is defined by 5 independent parameters:  $E = E_x = E_y$ ,  $E_z$ ,  $\nu_{xy}$ ,  $\nu_{zx}$ ,  $G = G_{zx} = G_{yz}$ :

$$\begin{bmatrix} \frac{1}{E} & -\frac{\nu_{xy}}{E} & -\frac{\nu_{zx}}{E_z} & 0 & 0 & 0 \\ -\frac{\nu_{xy}}{E} & \frac{1}{E} & -\frac{\nu_{zx}}{E_z} & 0 & 0 & 0 \\ -\frac{\nu_{zx}}{E_z} & -\frac{\nu_{zx}}{E_z} & \frac{1}{E_z} & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{2(1+\nu_{xy})}{E} & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{1}{G} & 0 \\ 0 & 0 & 0 & 0 & 0 & \frac{1}{G} \end{bmatrix}, \quad (7)$$

where  $E_i$  is the Young's modulus along axis  $i$ ,  $G_{ij}$  is the shear modulus in direction  $j$  on the plane whose normal is in direction  $i$ , and finally,  $\nu_{ij}$  is the Poisson's ratio defining the relationship between the transversal deformation (direction  $j$ ) and the axial elongation (when the loading is applied in direction  $i$ ).

The *contour structures* are modelled as an *isotropic* material behaviour, characterized by:

$$\begin{bmatrix} \frac{1}{E} & -\frac{\nu}{E} & -\frac{\nu}{E} & 0 & 0 & 0 \\ -\frac{\nu}{E} & \frac{1}{E} & -\frac{\nu}{E} & 0 & 0 & 0 \\ -\frac{\nu}{E} & -\frac{\nu}{E} & \frac{1}{E} & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{2(1+\nu)}{E} & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{2(1+\nu)}{E} & 0 \\ 0 & 0 & 0 & 0 & 0 & \frac{2(1+\nu)}{E} \end{bmatrix}, \quad (8)$$

The compliance tensor of isotropic materials depends on 2 independent parameters only, the Young's modulus and the Poisson's ratio.

### 3.3 THE REPRESENTATIVE VOLUME ELEMENT

Standard FE analysis of AM parts with inner structures such as in-fill or lattice structures is challenging because of the complexity of generating FE meshes suitable for such complex geometries. Generally, the FE discretization of in-fill structures as the one shown in Figure 1 implies a prohibitively high computational cost.

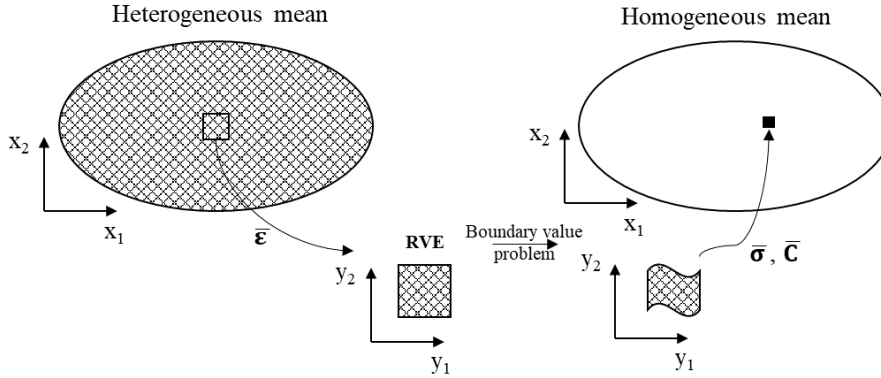
Let us consider two different scales of analysis: the global (macro) scale corresponding to the AM component and the local (micro) scale characterizing the inner structure accounting for both material and geometrical heterogeneities.

There are 3 main approaches to model heterogeneous structures:

1. The first approach is the *Direct Numerical Simulation* (DNS) where standard FE analysis is used, and all the information about the micro-structure is explicitly modelled [30]. This approach is efficient for small-scale analysis, but it soon becomes unaffordable for industrial studies because of the huge computational requirements in terms of both memory and CPU-time.

2. The second approach is known as *Concurrent Multiscale Method* (CMM), where a multiscale analysis involving the study of both length scales and the corresponding exchange of information among them is used. The micro-scale is solved at each sampling point of the structural model (macro-scale), establishing a strong coupling between macro and micro scales [31-32]. This method is also known as  $FE^2$  method because FE analysis is required at both scales.
3. The third approach is the *Computational Homogenization Method* (CHM). This approach establishes a weak coupling assuming a separation between the two length scales [33]. Therefore, the heterogeneities of the micro-scale are not directly inserted into the structural analysis but modelled through the definition of a *Representative Volume Element* (RVE). The RVE is then used to extract a structurally equivalent, homogeneous and generally anisotropic, constitutive behaviour to feed the macroscale at component level [34-36].

In this work, the homogenization strategy is adopted to replace the complex inner structure by a continuum (homogeneous) one, characterized by an equivalent structural response. This is feasible because of the *periodicity* of both in-fill and lattice structures defined by the repetition of a *unit cell*. Therefore, at global (component) level the volume is assumed to be anisotropic but homogeneous, while the heterogeneous internal geometry of in-fill or lattice structures is represented in the RVE (see Figure 12). Note that the main hypothesis defining the accuracy of the mechanical response of the RVE is based on the actual *scale separation*. Thus, the size of the printed part must be much larger than the RVE adopted for the homogenization procedure [37].



**Figure 12. Homogenization strategy**

### 3.4 HOMOGENEIZATION PROCEDURE

*First-order* homogenization is used in this work, where the strains are used as the link between the macro (component level) and the micro (inner structure) scales.

The problem at the microscale (RVE) is solved considering especial boundary conditions and then the homogenized variables are calculated through averaging equations [38].

The homogenization procedure consists of the following steps:

1. **Down-scaling.** The information is passed from the global (macro) scale to the local (micro) scale defined in terms of the RVE. Hence, the macroscopic strains, computed at the sampling points of the computational mesh used to discretize the AM component, are transferred to the RVE. According to the first-order homogenization technique, a *linear* displacement field is defined at each point  $\mathbf{X}$  in the macro-scale. Therefore, the macroscopic strains  $\boldsymbol{\varepsilon}_m(\mathbf{X})$  are *constant*. On the other hand, the displacement field at each point  $\mathbf{x}$  of the micro-scale (RVE) is much richer and cannot be solved by the coarse FE mesh used at the global level. The computational

homogenization method assumes that the macroscopic strains are obtained as the volume average of the microscopic strains,  $\boldsymbol{\varepsilon}_\mu(\mathbf{x})$  as:

$$\boldsymbol{\varepsilon}_m(\mathbf{X}) = \frac{1}{V_{RVE}} \int_{V_{RVE}} \boldsymbol{\varepsilon}_\mu(\mathbf{x}) dV \quad (9)$$

2. **Solution of the RVE problem.** Applying the Principle of Virtual Work, a Boundary Value Problem (BVP) defines the equilibrium of the RVE as:

$$\int_{V_{RVE}} d\boldsymbol{\varepsilon}_\mu(\mathbf{x}) \boldsymbol{\sigma}_\mu(\mathbf{x}) dV = F \quad (10)$$

where  $\boldsymbol{\sigma}_\mu(\mathbf{x}) = \mathbf{C}_\mu(\mathbf{x}) \boldsymbol{\varepsilon}_\mu(\mathbf{x})$  is the stress field at each point of the computational domain, being  $\mathbf{C}_\mu(\mathbf{x})$  the constitutive tensor.  $F$  is the work of the external loads, defined as:

$$F = \int_V \mathbf{v}_\mu(\mathbf{x}) \mathbf{b} dV - \int_V d\boldsymbol{\varepsilon}_\mu(\mathbf{x}) \mathbf{C}_\mu(\mathbf{x}) \boldsymbol{\varepsilon}_m dV \quad (11)$$

being  $\mathbf{v}_\mu(\mathbf{x})$  and  $d\boldsymbol{\varepsilon}_\mu(\mathbf{x})$  the virtual displacements and the virtual strains, respectively. Problem (11) must be complemented with the corresponding boundary conditions. In this work, the *Periodic Boundary Conditions* (PBC) are adopted for the solution of the BVP on the RVE:

$$\tilde{\mathbf{u}}(\mathbf{x}^+) = \tilde{\mathbf{u}}(\mathbf{x}^-) \quad \forall \{\mathbf{x}^+, \mathbf{x}^-\} \in \Gamma_{RVE} \quad (12)$$

where  $\mathbf{x}^+ \in \Gamma_{RVE}^+$  and  $\mathbf{x}^- \in \Gamma_{RVE}^-$  are *all* pairs of *opposite points* belonging to the RVE boundary  $\Gamma_{RVE} = \Gamma_{RVE}^+ \cup \Gamma_{RVE}^-$ .

This is the natural choice for both the in-fill and the lattice structures because of their actual periodicity. Moreover, their specific *unit cells* are selected as the optimal (most reduced) RVE domains.

3. **Up-scaling.** The solution obtained at local level is returned to the macro-scale through the homogenization process over the RVE. As a consequence of the Hill-Mandel Principle of Macro-Homogeneity [39-40], the stresses at the global level are obtained as the volume average of the local stresses computed on the RVE, as:

$$\boldsymbol{\sigma}_m(\mathbf{X}) = \frac{1}{V_{RVE}} \int_{V_{RVE}} \boldsymbol{\sigma}_\mu(\mathbf{x}) dV \quad (13)$$

The homogenized (*anisotropic*) constitutive tensor,  $\mathbf{C}_m$ , is obtained as:

$$\mathbf{C}_m = \mathbf{S} \cdot \mathbf{E}^{-1} \quad (14)$$

where  $\mathbf{S}$  is a matrix such that each column corresponds to the vector (Voigt's notation) of global stresses  $\boldsymbol{\sigma}_m^{(i)}$ , obtained through the homogenization process in Eq. 13, after solving the  $i^{th}$  BVP defined by Eq. 10. Thus:

$$\mathbf{S} = [\boldsymbol{\sigma}_m^{(1)} \quad \dots \quad \boldsymbol{\sigma}_m^{(6)}] = \begin{bmatrix} \sigma_{m,x}^{(1)} & \dots & \sigma_{m,x}^{(6)} \\ \vdots & \ddots & \vdots \\ \sigma_{m,yz}^{(1)} & \dots & \sigma_{m,yz}^{(6)} \end{bmatrix} \quad (15)$$

Therefore, each BVP is characterized by a loading condition (3 elongations and 3 distortions) defined in terms of the corresponding prescribed strain field:  $\boldsymbol{\varepsilon}_m^{(i)}$ ,  $i = [1,6]$ . Hence, the matrix  $\mathbf{E}$  is filled with the components of the global strains,  $\boldsymbol{\varepsilon}_m^{(i)}$ , as:

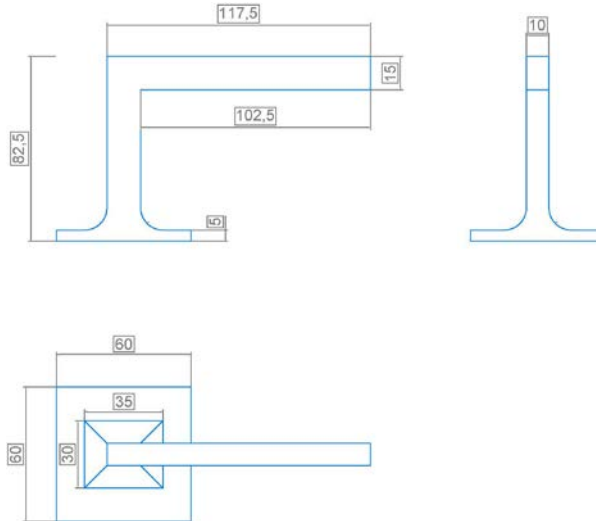
$$\mathbf{E} = [\boldsymbol{\varepsilon}_m^{(1)} \quad \dots \quad \boldsymbol{\varepsilon}_m^{(6)}] = \begin{bmatrix} \varepsilon_{m,x}^{(1)} & \dots & \varepsilon_{m,x}^{(6)} \\ \vdots & \ddots & \vdots \\ \varepsilon_{m,yz}^{(1)} & \dots & \varepsilon_{m,yz}^{(6)} \end{bmatrix} \quad (16)$$

### 3.5 CALIBRATION OF THE NUMERICAL MODEL

In this Section a door-handle demonstrator is used for the calibration of the numerical model developed. A sensitivity analysis is carried out to calibrate the material properties of both the in-fill and the contour structures by comparison with the experimental measurements.

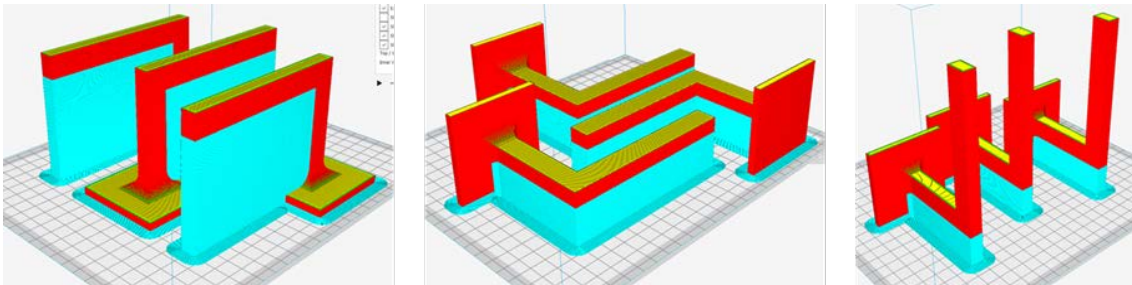
The door-handle components are fabricated using the same material (PLA) and the same FFF printer as the ones used for the experimental campaign on dog-bone samples.

The dimensions of the component are illustrated in Figure 13. Three different orientations are considered in the fabrication as shown in Figures 14 and 15. The numerical model is also expected to predict the structural performance of FFF components according to the printing orientation chosen.

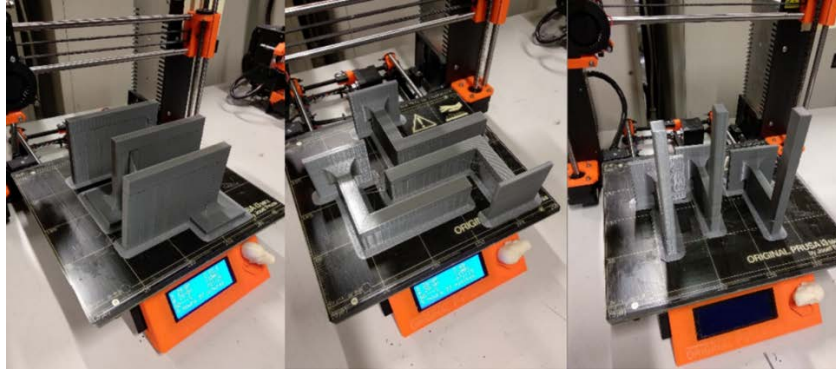


**Figure 13. Dimensions of the door handle tested**

The supporting structures (breakaway type) shown in blue in Figure 14 are fabricated using the same material as the component. Such structures are generally required when the external surfaces exhibit an angle lower than  $30^\circ$  with respect to the horizontal (XY) plane.



**Figure 14. Door handle demonstrator: orientation 1, 2, 3 (left to right)**



**Figure 15. Door handle samples printed in orientation 1, 2, 3 (left to right)**

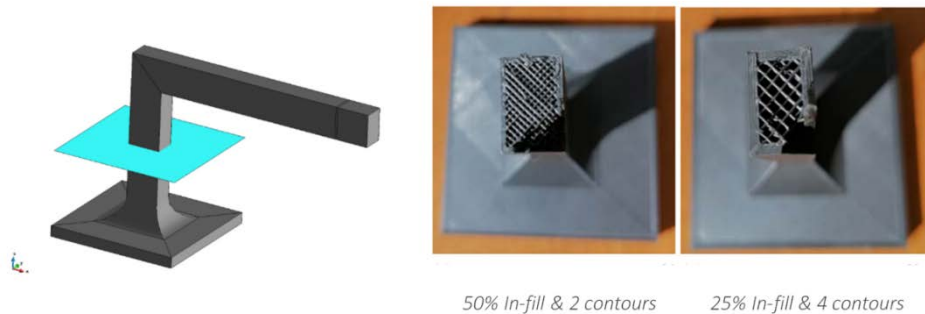
The door-handle components are divided into 4 families. Each family is characterized by 2 different in-fill densities (25% and 50%, respectively) and 2 types of contours: single or double contour thickness. Table 4 summarizes the different printing parameters that characterize the 4 families.

**Table 4. Door handle printing information**

<i>In-fill density (%)</i>	50	25
<i>N° filaments × contour</i>	4 (double contour)	2 (single contour)
<i>Printing orientation</i>	3 orientations	
<i>Material</i>	PLA	

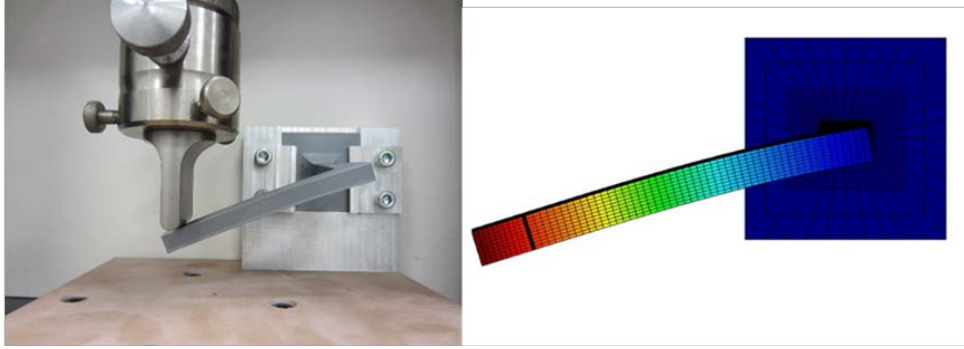
The different printing patterns and orientations are generated using the Cura software by Ultimaker. The in-fill is characterized by  $+45^\circ / -45^\circ$  alternating raster configuration [27, 41].

A cross section of the demonstrator printed in orientation 1 is shown in Figure 16. Both in-fill densities, as well as, the 2 different contour thicknesses of the external contour can be clearly appreciated.



**Figure 16. Door handle printed in orientation 1. Cross section characterized by 50% in-fill and a single contour (left); 25% in-fill and double contour (right).**

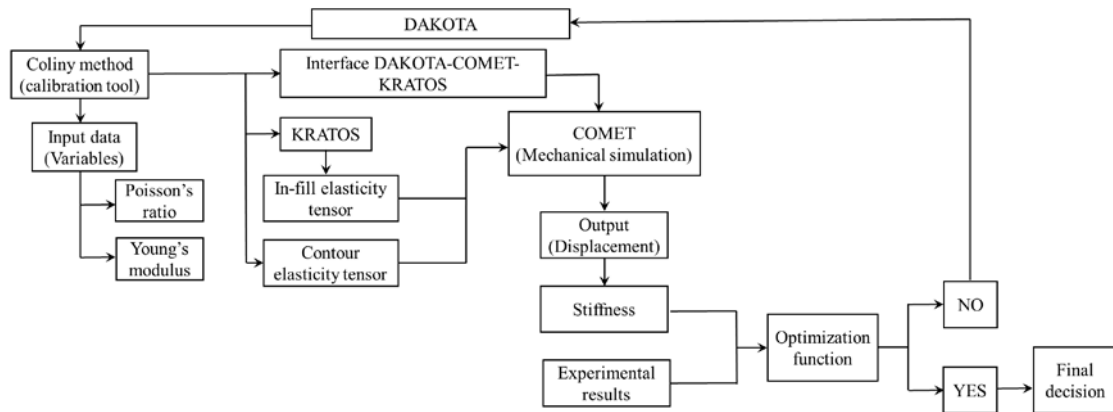
The loading consists of an imposed vertical displacement of 35 mm applied at 15 mm from the end of the door-handle and applied at a loading speed of 1 mm/min. Thus, a combination of bending and torsion characterizes the resulting stress state as shown in Figure 17.



**Figure 17. Experimental set-up and load location (left) Numerical result (right)**

The mechanical performance is analysed for each orientation and any combination of the printing parameters under the same loading conditions. The load vs. displacement curve is recorded for each test. In this work, only the linear elastic behaviour, characterized by the structural stiffness of the demonstrator is analysed.

The calibration of the mechanical model combines the use of the several software modules. Figure 18 shows the flow chart of the numerical procedure used.



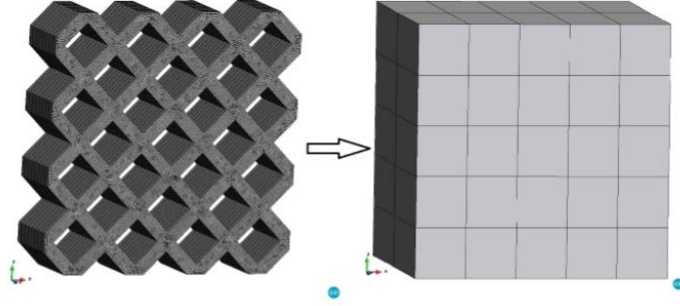
**Figure 18. Flow chart of the iterative process used for the numerical calibration**

In the present work, a specific KRATOS module [42] has been implemented to extract the constitutive tensor of the actual in-fill or lattice structure (see Figure 19). Periodic boundary conditions are applied to the RVE in the homogenization procedure. A sensitivity analysis on the number of unit cells of the original in-fill structure is performed as shown in Figure 20. It is found out that, adopting periodic boundary conditions, the mechanical response is not sensitive to the number of unit cells. Therefore, the size of the RVE can be reduced to the size of one unit cell minimizing the computational cost.

It must be noted that the anisotropic behaviour of the in-fill (or lattice) structures depends only on the geometrical features that characterize the unit cell. Thus, the same properties of the printed material adopted for the external contours are used to characterize the bulk material of the in-fill structure.

This homogenization module is used to generate an *off-line* data-base including the most typical in-fill and lattice structures used in FFF.

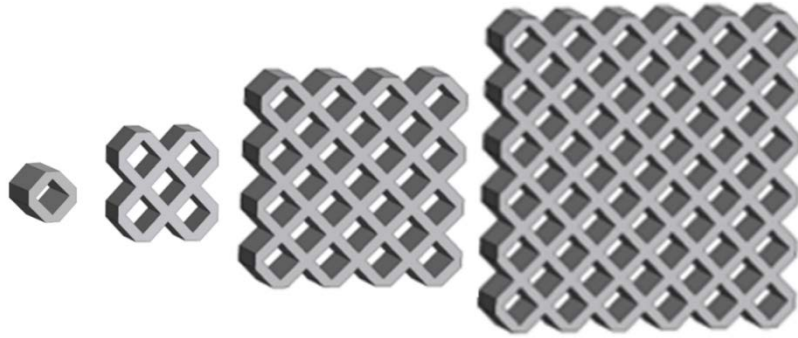




**Figure 19. Homogenization procedure using RVE: from the heterogeneous in-fill structure (left) to the homogeneous but anisotropic counterpart (right)**

The overall structural behaviour according to the mechanical loading and boundary conditions applied is computed using COMET, a finite element based in-house software [43].

As discussed before, the volume of the component to be analysed is split into two computational domains: the contour and in-fill structure, respectively. The contour is assumed to be homogeneous and isotropic while the in-fill (or lattice) structure is replaced by the equivalent homogeneous but anisotropic constitutive tensor computed by the KRATOS homogenization module.



**Figure 20. Several RVE domains used for the sensitivity analysis**

In this work, DAKOTA, a well-known open source optimization tool developed by Sandia National Labs [44], is used to calibrate the numerical model and, in particular, to minimize the difference between the structural stiffness as experimentally measured, and its numerical counterpart. The procedure is repeated for the 3 orientations and 2 set of printing parameters: (i) single contour (2 filaments) and 50% in-fill; (ii) double contour (4 filaments) and 25% in-fill.

The initial guess of the calibration loop is taken from the values of the Young's modulus and the Poisson's ratio as proposed by the filament manufacturer. The admissible range is defined according to the values obtained through the experimental campaign for Family-2. Therefore, the Young's modulus is allowed to vary between 2700 to 3200 MPa and the Poisson's ratio between 0.18 to 0.25. Coliny *derivative-free* method is chosen to find their optimal values minimizing the objective function,  $S$  defined as:

$$S = \sum_{i=1}^n r_i^2, \quad (17)$$

where  $r_i = K_{exp,i} - K_{sim,i}$ , are the residual errors between the computed and the experimental values of the structural stiffness,  $K_{sim,i}$  and  $K_{exp,i}$ , respectively. These values

are computed as the slope of the load vs. displacement curves according to each printing configuration (*i*), defined in terms of both orientation and process parameters setup.

### 3.5.1 RESULTS

In this section, the mechanical properties obtained through the calibration procedure for both contour and in-fill materials are shown in Tables 5 and 6, respectively. Comparing these values with the material properties of the raw material, a difference of about 20% in Young's modulus and 30% in the Poisson's ratio is observed. The properties of the raw material result to be stiffer than the ones of the printed components.

**Table 5. Optimized values for the contour**

Mechanical properties	Contour
Young's modulus (MPa)	2700
Poisson's ratio	0.25

**Table 6. Transversally isotropic properties for the homogenized in-fill**

%In-fill		25%	50%
Young's modulus (MPa)	$E_x$	15.494	179.3
	$E_y$	15.494	179.24
	$E_z$	674.96	1350.0
Shear modulus (MPa)	$G_{xy}$	186.71	393.21
	$G_{yz}$	150.39	344.97
	$G_{xz}$	150.39	344.98
Poisson's ratio	$\nu_{xy}$	0.9599	0.8049
	$\nu_{yz}$	0.0057	0.0332
	$\nu_{xz}$	0.0057	0.0332

Table 7 shows the relative error between the structural stiffness as measured experimentally and by numerical simulation for each printing orientation and set of process parameters. The relative error is below 7% for most of the cases; thus, a good calibration of the demonstrator is achieved.

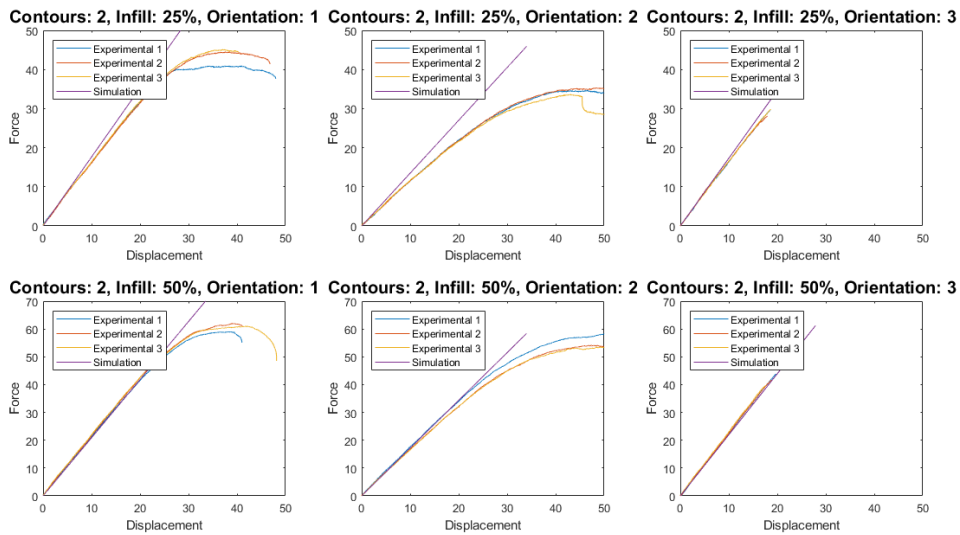
The load vs. displacement curves obtained for all orientations are shown in Figure 21 and Figure 22 for the cases of 4 filaments and 2 filaments through the contour thickness, respectively. A very good agreement between the numerical simulation and the experimental results is obtained.

As expected, the highest structural stiffness is shown by the family printed with the thickest external contour (4 filaments) and the densest in-fill structure (50%). Contrarily, the most flexible configuration corresponds to 2 filaments through the contour and 25% in-fill. Therefore, it is also possible to conclude that the structural stiffness generally depends on the size of the external contour, while the in-fill has a minor role.

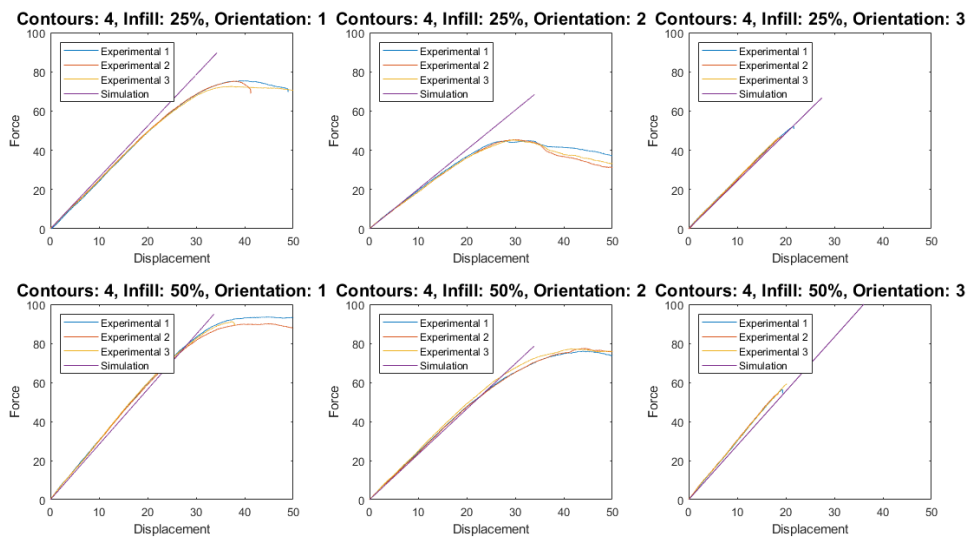
For each family, orientation 2 is more flexible than the other two orientations, while the orientations 1 and 3 show very similar structural behaviour. Moreover, the numerical results corresponding to this weaker orientation, and particularly when using the lowest in-fill density and contour thickness, exhibits the worst numerical agreement because the original hypothesis of linear elastic response is no longer fulfilled, as clearly shown by the load vs. displacement curves (see Figure 21). Contrarily, for stiffer solutions the relative error reduces to 3,75% (Figure 23).

**Table 7. Best numerical approximation of the mechanical response of the door-handle and the relative error according to each orientation and printing parameters**

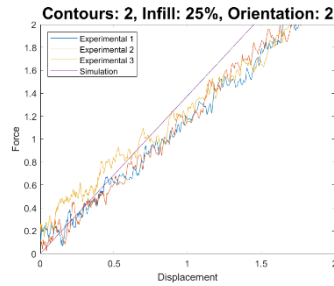
Orientation	# Filaments	In-fill %	Displacement (mm)	$K_{\text{Numerical}}$ (MPa)	Force (N)	$K_{\text{Experimental}}$ (MPa)	$\epsilon_{\text{rel}} \%$	$r_1^2$
1	4	25	5.73643	2.615	15	2.4435	7	0.029366793
2			7.44865	2.014		1.8769	7	0.018738037
3			6.16746	2.432		2.5339	4	0.010359478
1	4	50	5.32085	2.819	15	3.0005	6	0.032907137
2			6.46861	2.319		2.4316	5	0.012703
3			5.40936	2.773		3.0127	8	0.057469164
1	2	25	8.50330	1.764	15	1.5999	10	0.026935963
2			11.11501	1.350		1.109	22	0.057852974
3			8.66519	1.731		1.6425	5	0.007843543
1	2	50	7.17343	2.091	15	2.1333	2	0.001785185
2			8.73482	1.717		1.6751	3	0.001777926
3			6.83087	2.196		2.2664	3	0.004968353
							S	0.262707552



**Figure 21. Force [N] vs. displacement [mm] curves comparing the simulation results with the experimental measurements (single contours).**



**Figure 22. Force [N] vs. displacement [mm] curves comparing the simulation results with the experimental measurements (double contours).**



**Figure 23. Force [N] vs. displacement [mm] curves for the simulation results and the experimental tests (2 contours and 25% in-fill and orientation 2)**

From the analysis of the demonstrator, it can be concluded that the strategy adopted for the numerical simulation of structural behaviour of AM components built by FFF is adequate as long as the hypotheses (e.g. linear elasticity, small strains, scales separation for in-fill structure homogenization, etc...) apply. Moreover, the numerical simulations performed confirmed that modelling the contour of the door handle as isotropic is an acceptable choice.

#### 4 DISCUSSION AND FUTURE LINES

The experimental campaign carried out in this work supports that FFF induces anisotropic behaviour of the built parts, enhancing the relevance of the printing orientation as the most important issue to be considered to optimize the structural behaviour.

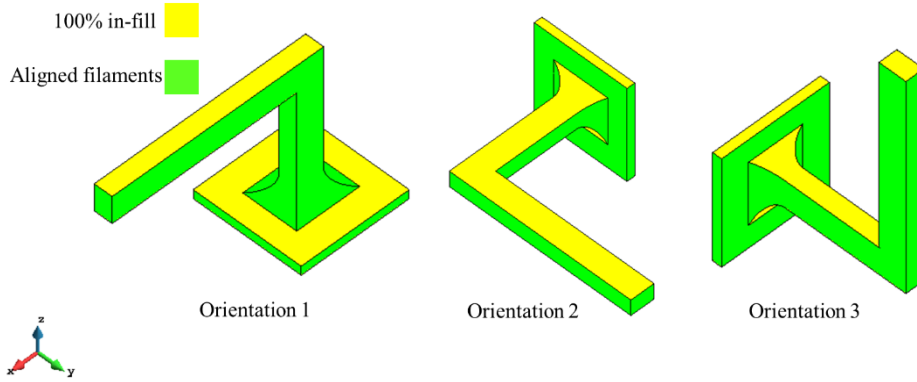
The printing patterns chosen by the most common slicer tools (e.g. Cura, Magic, etc...) typically vary as a function of the slope of the external contour with respect to the built plane. On the one hand, when the angle between the built plate and the contour is higher than  $30^\circ$ , the filaments are aligned with the contour orientation and piled up according to the layer-by-layer AM strategy. On the other hand, when this angle is lower than  $30^\circ$  and, most typically, for horizontal contours (covers, etc..) the crossed  $+45^\circ$   $-45^\circ$  patterns is usually preferred. We will refer to the latter as 100% in-fill because the raster distribution is the same as used for the fabrication of the in-fill structures but, in this case, the entire volume is (100%) filled by the extruded material.

Figure 24 shows both contours for each orientation where the yellow and green colours are the 100% in-fill and the contour with aligned filaments, respectively.

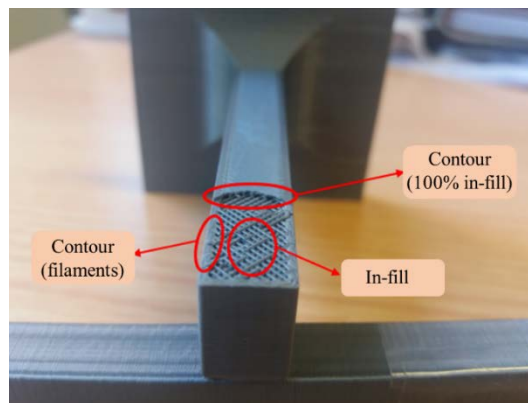
The contour surfaces parallel to the printing plane (XY) are made of 100% in-fill with raster distribution, while the contour surfaces growing in the vertical direction (Z) are made of aligned filaments.

Figure 25 illustrates the different parts of the door handle component including in-fill and contours of 100% in-fill and filaments.

Contours printed with aligned filaments present a different mechanical behaviour, in terms of both stiffness and strength, compared to those made by 100% in-fill. This is mostly due to the adhesion among filaments. Thus, adhesion plays a key role in FFF, being the contact surface among filaments related to the selected printing pattern. This supports further investigation of the sensitivity of the mechanical behaviour with respect to the selected printing patterns. An extended experimental campaign is required to perform tensile tests on FFF specimens with different printing patterns used.



**Figure 24. Three printing orientations considered for the door handle: 100% in-fill contours (yellow); contours with aligned filaments (green).**



**Figure 25. Different printing patterns of the actual door handle component**

## 5 SUMMARY AND CONCLUSIONS

In this work, the mechanical performance of the AM components made by FFF is analysed experimentally and numerically. The geometry of printed parts is split into two parts: the external contours and the inner structure (in-fill or lattice structures). The material of the external contour surfaces is characterized experimentally by tensile tests on printed dog-bone specimens. It is observed that the material parameters are about 20-30% lower compared to that of the raw material. Moreover, the mechanical behaviour of the aligned-type and crossed-type samples are transversely isotropic. Depending on the printing pattern, the isotropic plane varies. The crossed-type samples are transversely isotropic where  $XZ$  is the isotropic plane with the stiffness lower in  $Y$  direction. The mixed-type and aligned-type samples are also transversely isotropic. However,  $YZ$  is the isotropic plane and the stiffness is lower in  $X$  direction.

A homogenization technique is applied to the inner structures to obtain the equivalent homogeneous but anisotropic counterpart to be used for the numerical analysis of both in-fill and lattice structures. Periodic Boundary Conditions (PCB) are applied to the selected RVE to obtain the required constitutive tensor. This strategy is repeated *off-line* to generate a data-base for the most common inner structures identified according to their printing patterns and raster densities.

Different software modules are developed to conform the numerical platform used for the numerical simulation: Kratos, an in-house multi-physics software, is used to characterize the inner structures by RVE homogenization techniques; Comet, another in-house FE

solver, is used for the overall structural analysis of AM components. Finally, Dakota, an optimization software, is used as to drive the optimization loop used for the calibration procedure. The optimal material parameters are obtained by minimizing the difference between the numerical results and the experimental measurements.

The door-handle components printed along 3 different orientations and adopting several printing parameters are used as demonstrators. The numerical simulations performed confirmed that modelling the contour of the door handle as isotropic is an acceptable choice.

The good agreement achieved validates the numerical strategy adopted, and, therefore, the predictive capability of the proposed numerical model.

Moreover, the correlation between the printing orientation and the structural performance of FFF components is demonstrated. This shows that it is important to select the optimal printing orientation as well as the most suitable contour thickness and in-fill density to satisfy the structural performance required.

It is also shown that the mechanical properties of the inner structures are much lower compared to the stiffness of the external contour. However, the use of these inner structures improves the structural performance of a FFF part enhancing structural stability of the overall component according to their raster density and printing orientation.

Overall, the relevance of the present work is that it raises the topic of distinguishing between the mechanical behaviour of the contour and the in-fill structures, and the determination of their respective mechanical properties. On the one hand, performing experiments to characterize the contour and the in-fill is a challenging task. On the other hand, it is possible to calibrate the numerical model through a sensitivity analysis based on the as-printed elastic parameters. Moreover, it is shown that assuming an isotropic behaviour for the external contour results in a negligible error.

## **ACKNOWLEDGMENT**

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